

ANK 315 AIRCRAFT MAINTENANCE

INTRODUCTION TO DIAGNOSTICS NDT, SHM

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Projekt "Program rozwoju dydaktycznego Wydziału Mechanicznego Energetyki i Lotnictwa



NON-DESTRUCTIVE TESTING

The need of non-destructive testing (NDT) application in aviation demands much of the equipment and technique makers qualification what is connected with a great variety of shape and complexity of testing units

Thereby, it is usually necessary to test the areas of stress concentration: holes for fasteners, panels edge, fillet transitions etc. In operation it is often essential to carry out NDT without removing the layer of paint-and-lacquer coating or sealant what excludes the usage of penetrant and ultrasonic techniques. In many cases there is no two-way access what makes it impossible to apply X-ray NDT technique

Moreover, it is often necessary to perform NDT in field conditions and airfield conditions During field service of aircrafts an operator with the equipment should be located directly on the wing or stepladder

Basic Technology Requirements Range

- Detection capability
- Temperature loading
- Chemical loading
- Mechanical loading
- Manufacturing aspect
- System integration aspect
- Maintainability
- Reparability
- Self-diagnostic capability

Reliability > 30 years

Non Destructive Testing NDT NDE NDI

- continuous damage monitoring
- Structural Health Monitoring (SHM)
- Health and Usage Monitoring System (HUMS)
- Engine Monitoring System (EMS)

NON-DESTRUCTIVE TESTING

NDT TECHNIQUES

OBSERVE AND DIRECTLY DEFECT DAMAGE

VISUAL/OPTICAL, ULTRASONICS, SHEROGRAPHY, THERMOGRAPHY, etc.

STRUCTURAL HEALTH MONITORING

DETECT AND MONITOR USING INDIRECT MEASUREMENTS

VIBRATION/MODAL ANALYSIS, ACUSTO-ULTRASONICS, FIBER BRAGGS GRITINGS etc.

- Visual, Optical
- Liquid Penetrant
- Magnetic
- Eddy Current
- Radiography (X-ray/ gama ray)
- Sonic/Resonance
- Ultrasonic
- Infrared Thermography
- Sherography

- Visual, Optical
- Liquid Penetrant
- Magnetic
- Eddy Current
- Radiography (X-ray/ gama ray)
- Sonic/Resonance
- Ultrasonic
- Infrared Thermography
- Spectography

VISUAL INSPECTION

 ALL HUMAN SENCES INVOLVED
 HIGHLY DEPENDENT ON CAPABILITIES OF OBSERVER, WHO IS TRAINED TO FIND AND ASSESS CHANGES IDENTIFING DAMAGES

THERE ARE TWO GROUPS:

 1ST IS DIRECT VISUAL INSPECTION USE BOROSCOPES, FIBEROSCOPES WITH REAL-TIME VIDEO FOR DIRECT OBSERVATION

2ND IS INDIRECT VISUAL INSPECTION BASED ON EXAMINATION OF PHOTOGRAPHS, RADIOGRAPHS OR VIDEOTAPES



Drobne wyposażenie do badań wizualnych: a) lupy, mikroskopy piórowe (u dołu) i mikroskopy pomiarowe (w środku), b) skale do lup i mikroskopów do badań wizualnych, c) lupa z oświetlaczem i skalą, firmy Heine. Za zgodą firmy Heine















VISUAL INSPECTION

- VISUAL AIDS SUCH AS LIGHT SOURCES, MAGNIFICATION DEVICES AND SENCORS FOR REMOTE VIEWING EMPLOYED
 - UP TO 80% OF CRACKS ARE FOUND BY VISUAL MEANS
 - INEXPENCIVE, HIGHLY PORTABLE, IMMEDIATE RESULTS, MINIMAL TRAINING
 - APPROPRIATE FOR ONLY SURFACE DEFECTS, DAMAGES
 - HUMAN FACTOR

VISUAL INSPECTION

INEXTENSIVE, HIGHLY PORTABLE, IMMEDIATE RESULTS, MINIMAL TRANING

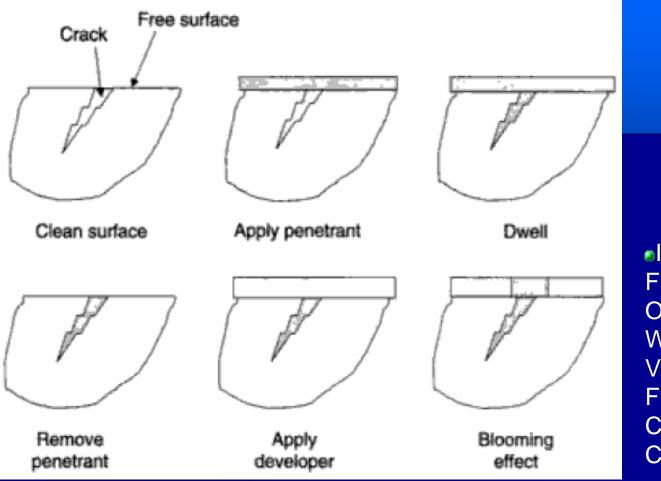
ONLY SURFACE DEFECTS, CLEANING REQUIRED, VISUAL ACCESS REQUIRED, HUMAN FACTOR

- Visual, Optical
- Liquid Penetrant
- Magnetic
- Eddy Current
- Radiography (X-ray/ gama ray)
- Sonic/Resonance
- Ultrasonic
- Infrared Thermography
- Sherography

LIQUID PENETRANT INSPECTION

- RELATIVELY SIMPLE BUT EFFECTIVE TECHNIQUE FOR LOCATING SURFACE CRACKS
- APPLICABLE TO MANY MATERIALS



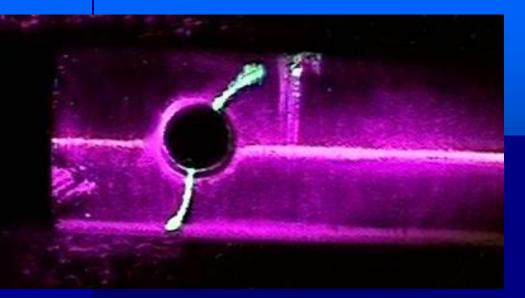


INVOLVES
 FLOODING SURFACE
 OF TEST OBJECT
 WITH LOW VISCOSITY FLUID.
 FLUID SOAKS INTO
 CREVICES VIA
 CAPILLARY ACTION

PENETRANT IS CLEANED AND DEVELOPER IS APPLIED TO DRAW OUT PENETRANT REMANING BETWEEN CRACK FACES

 DEVELOPER PROVIDES CONTRASTING BACKGROUND AND BLOOMING EFFECT TO ENHANCE CRACK VISIBILITY

Liquid penetrant inspection





PENETRANTS CAN BE FLUORESCENT AND VIEWED UNDER SUITABLE e.g. ULTRAVIOLET LIGHT



LIQUID PENETRANT INSPECTION

PORTABLE AND WELL SUITED FOR FIELDWORK. INEXPENCIVE AND REQUIRES MINIMAL SKILLS. SENSITIVE TO SMALL DISCONTINUITIES. CAN BE APPLIED TO NONPOROUS MATERIALS. FLAW ORIENTATION DOES NOT USUALLY POSE A PROBLEM

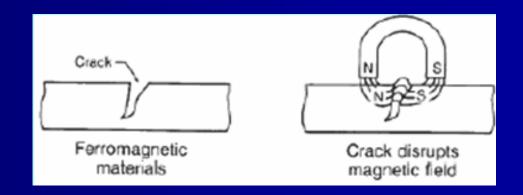
ONLY SURFACE ANOMALIES. MUST HAVE ACCESS TO SURFACE. STRESSES AT SURFACE CAN CAUSE UNCERTAINLITY. SIGNIFICANT SURFACE PREPARATION REQUIRED. FAIRLY SLOW DUE TO PREPARATION, DWELL AND CLEAN UP

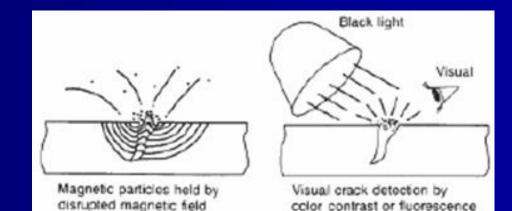
- Visual, Optical
- Liquid Penetrant
- Magnetic
- Eddy Current
- Radiography (X-ray/ gama ray)
- Sonic/Resonance
- Ultrasonic
- Infrared Thermography
- Sherography

Magnetic Particle Inspection

FERROMAGNETIC TEST PIECE IS MAGNETISED. MAGNETIC FLUX LEAKAGE FIELDS ARE DETECTED THROUGH THE ADDITIONAL OF SMALL PARTICLES WHICH MIGRATE TO THE FLUX LEAKAGE

CAN DETECT SURFACE AND NEAR SURFACE FLAWS

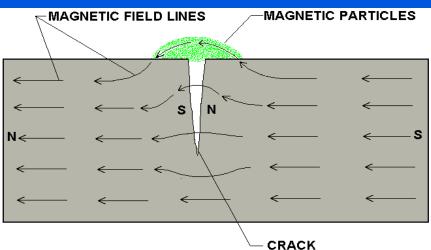




in "black light

Magnetic Particle Inspection



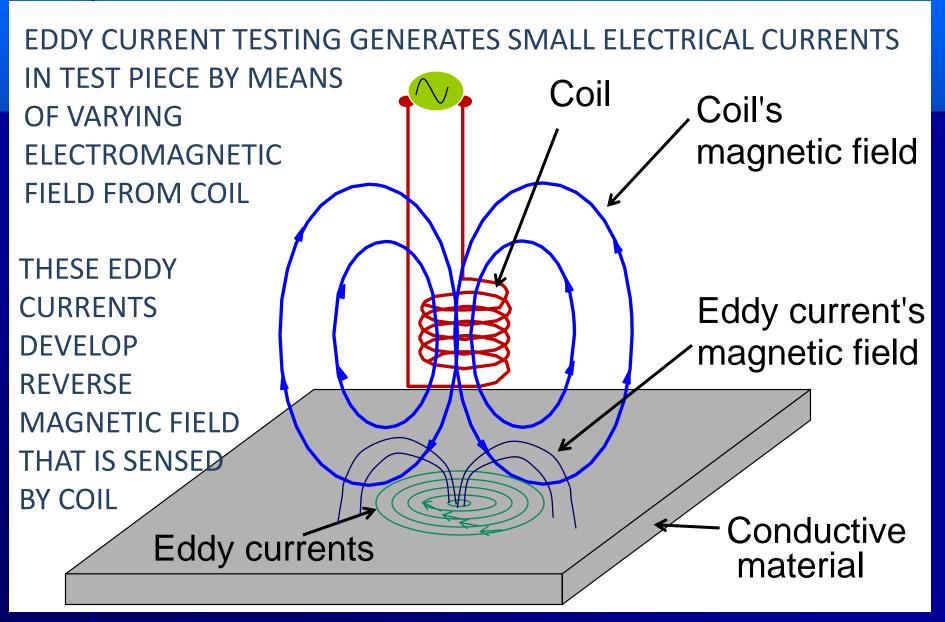


Magnetic Particle Inspection

COMPLEX SHAPES READILY TESTED. FAIRLY PORTABLE AND RELATIVELY INEXPENSIVE. SENSITIVE TO SMALL SURFACE(OR NEAR-SURFACE) DFECTS. REQUIRES MODERATE SKILL LEVEL. RELATIVELY QUIK RESULTS

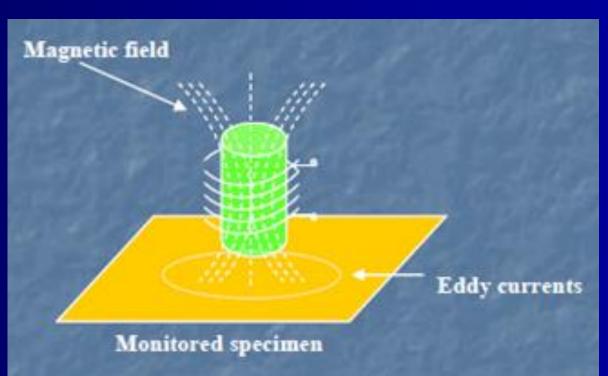
LIMITED TO FERROMAGNETIC MATERIALS. SURFACES MUST BE ACCESIBLE. LARGE ELECTRIC CURRENT CAN BE REQUIRED. MAY NEED TO MAGNETIZE OBJECT. LIMITED TO SURFACE AND NEAR SURFACE DEFECTS. SURFACE FINISHES MAY NEED TO BE REMOVED. FLAW DETECTION SENSITIVE TO MAGNETIC FIELD ORIENTATION

- Visual, Optical
- Liquid Penetrant
- Magnetic
- Eddy Current
- Radiography (X-ray/ gama ray)
- Sonic/Resonance
- Ultrasonic
- Infrared Thermography
- Sherography



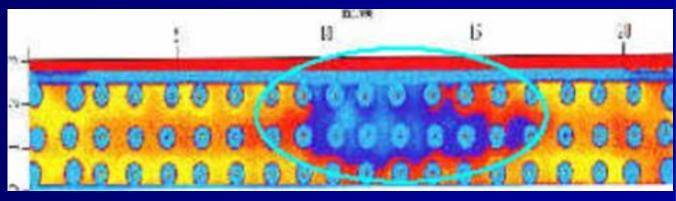
ANYTHING THAT AFFECTS EDDY CURRENTS (E.G. CRACKS, POROSITY, INCLUSIONS) WILL MODIFY SECONDARY MAGNETIC FIELD

AND BE SENSED BY IMPEDANCE CHANGE IN COIL



Edddy current inspection is a surface method and can only detect defectts up to a depth of around 6 mm
 The method requires calibration(artificial defects such as saw cuts, holes etc.)



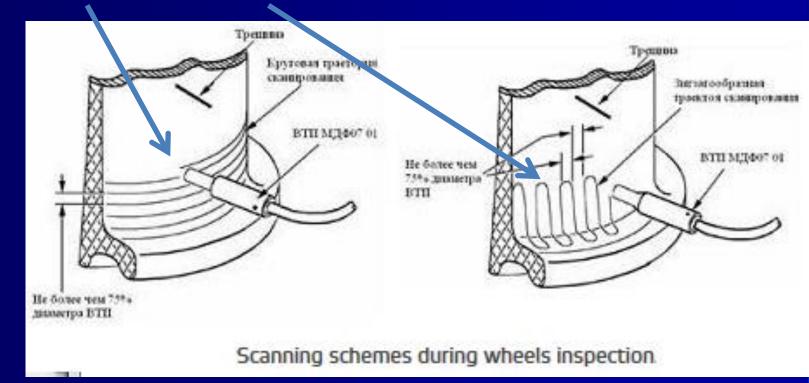


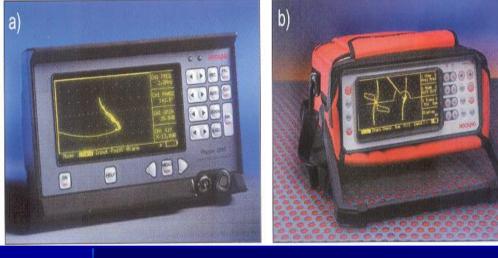
Three probe types:
Surface probes for testing plates
Encircling probes for bars and rods
Bobbin probes for inspection of tubes

Practicing of local corrosion damages detection technique



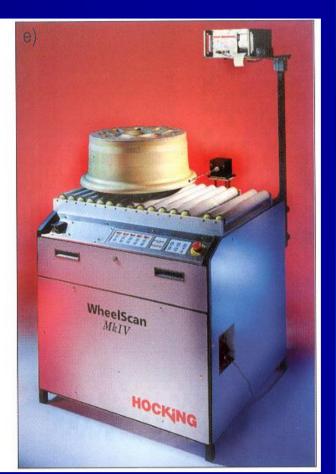
Two scanning types are applied: circular and zigzag









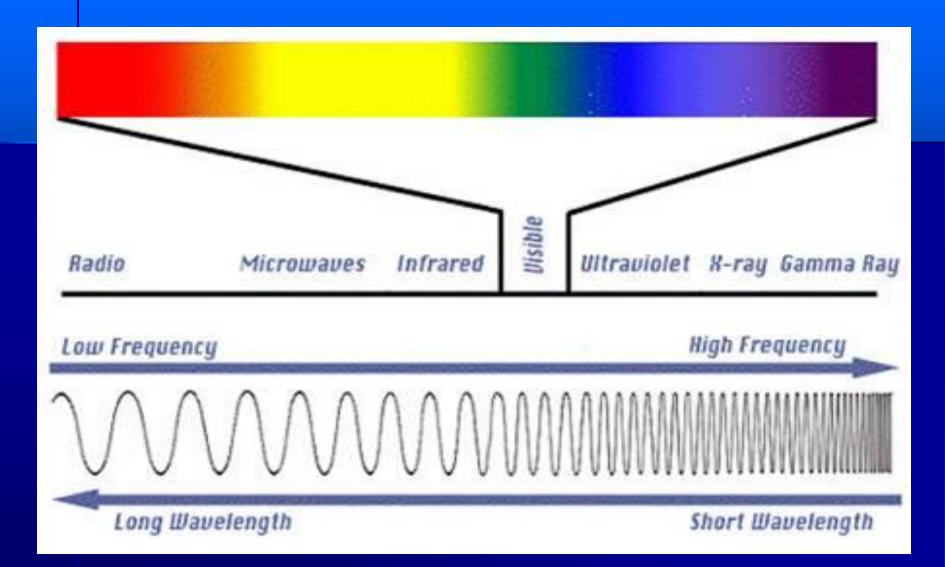




SENSITIVE TO SEVERAL STRUCTURAL AND METALLURGICAL VARIABLES LEADING TO MANY APPLICATIONS. MODERATELY FAST TECHNIQUE GIVING INSTANTANEOUS INDICATION. PORTABLE. CAN DETECT SMALL SURFACE OR NEAR-SURFACE ANOMALIES. MINIMAL PART PREPARATION

TEST OBJECT MUST BE ELECTRICAL CONDUCTOR. SURFACE MUST BE ACCESSIBLE TO PROBE. LIMITED DEPTH OF PENETRATION. SKILL AND TRAINING REQUIRED TO INTERPRET RESULTS. CAN BE TIME CONSUMING FOR LARGE INSPECTION AREAS

- Visual, Optical
- Liquid Penetrant
- Magnetic
- Eddy Current
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- Sonic/Resonance
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- Sherography



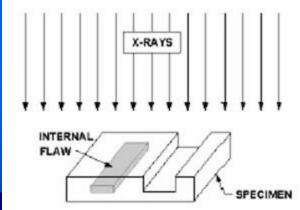
RADIOGRAPHIC INSPECTION

RADIOGRAPHY IS ONE OF THE OLDEST NDT METHODS FOR LOCATING INTERNAL AND EXTERNAL ANOMALIES

EXPLOITS CHARACTERISTIC OF MANY MATERIALSTO DIFFERENTIALLY ABSORB AND PASS HIGH-ENERGY RADIATION

SOURCE OF RADIATION MAY BE X-RAYS, GAMMA RAYS OR NEUTRONS

VARIATIONS IN ABSORPTION WHICH RESULT FROM CHANGES IN MATERIAL DENSITY OR THICKNESS ARE RECORDERED USING EITHER FILM OR REAL-TIME DETECTORS





RADIOGRAPHIC INSPECTION

APPLICATIONS INCLUDE:

CHECKING FOR POROSITY IN CASTINGS OR WELD DEFECTS DETERMINING INTERNAL GEOMETRY DETECTING CRACKS IF PROPERLY ORIENTED LOCATING AND MEASURING CORROSION THINNING





ROBOSCAN 2M AERIA

Radiographic Inspection

CAN DETECT INTERNAL AND EXTERNAL DEFECTS. APPLICABLE TO WIDE VARIETY OF SOLID MATERIALS AND COMPLEX SHAPES. PERMAMENT RECORD OBTAINED. MINIMAL PART PREPARATION. REAL-TIME VIEWING POSSIBLE. 3D IMAGING POSSIBLE

SIGNIFICANT SAFETY ISSUES. TWO-SIDES ACCESS TO TEST PIECE GENERALLY REQUIRED. DIRECTION VERY SENSITIVE TO FLAW ORIENTATION. HIGH DEGREE OF SKILL AND EXPERIENCE. RELATIVELY EXPENSIVE. USUALLY SOME DELAY IN OBTAINING RESULTS

9 NDT TECHNICS ARE USED IN AVIATION

- Visual, Optical
- Liquid Penetrant
- Magnetic
- Eddy Current
- Radiography (X-ray/ gama ray)
- Sonic/Resonance
- Ultrasonic
- Infrared Thermography
- Sherography

Acoustic Emission (AE) has been defined as "a transient elastic stress wave generated by the rapid release of energy from a localized source within a material" – 1995 Annual Book of American Society of Testing and Materials Standards, Nondestructive Testing.

Most non-destructive evaluation applications involve loading test object, sensing AE and correlating it with source of damage.

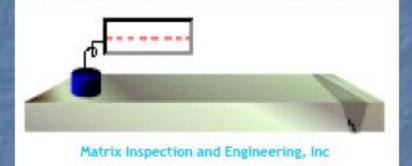
Modern utilisation of AE began in 1950s and has success with "noisy" materials such as composites and concrete.

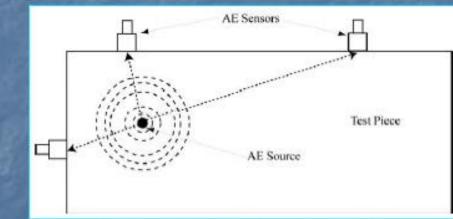
 Many sources of AE including crack formation and growth and corrosion activity.

 Sources of AE in composite materials include fibre breakage and delamination.

AE involves "listening" for sound generated by growing damage in the test piece.

Location of AE damage can be determined by triangulation techniques based on time for signal to reach sensors – damage size cannot be found by AE.





- Each loading event emits unique burst of sound so AE inspections not directly repeatable.
- Although AEs generated by material and not external source, stimulus is required to trigger original AEs – can be mechanical, thermal or magnetic induced stress.
- Although AE signals cover wide range of frequencies and energy levels, they consist of two basic types:

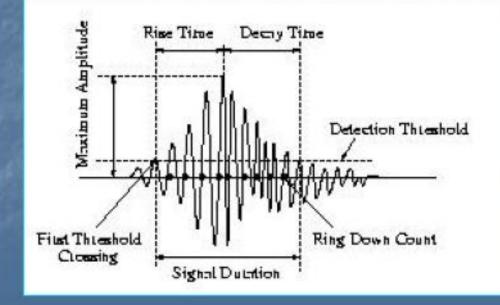
Although AE signals cover wide range of frequencies (typically 20kHz – 1200kHz) and energy levels, they consist of two basic types:

- Burst Emissions correspond to particular emission event such as development of cracks in metals.
- Continuous Emissions give sustained signal associated with rapidly occurring events such as motion of dislocations.

Both types of emissions may propagate by combinations of standard ultrasonic modes.

Early AE applications gave inconsistent or misleading results due to inadequate standards.

Problem addressed in recent years with generally accepted AE parameters:



Acoustic Emission are ultrasonic waves and are measured using similar methods:

 Piezoelectric transducers – most common – sensitive and reliable and suitable for wide range of frequencies.

 Accelerometers – limited frequency range (100kHz maximum).

 Optical/laser transducers – employ laser interferometry – wide frequency response and ability to integrate signals over large area.





AE signals locate source of growing damage in loaded structure. Monitoring equipment relatively simple and expensive. Signals can be monitored online or recorded and analysed later. Can detect discontinuities below surface. Wide-area coverage possible.

Applied stress or chemical activity required for emissions to occur – stabilised damage not detect. Not all damage emits AE. Not applicable to all materials. Crack sizes not readily determined. Method not directly repeatable. Multiple sound paths can complicate signal identification.

Sonic/Resonance

Resonance test method
Pitch/catch swept test method
Pitch/catch impulse test method
MIA(Mechanical Impedance Analysis) test method
Eddy sonic harmonic test method
Tap test

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Ultrasonic inspections measure travel of highfrequency sound waves introduced into test object at various surface locations.

Propagation of these electronically controlled sound pulses through test object is detected at specific points where acoustic energy is converted back to electrical signal.

Interpretation of reflection and refraction of sound waves at boundaries leads to information regarding discontinuities in test object.

Since sound waves travel in many materials, ultrasonic inspection is one of most widely used NDT tool for internal and surface flaws.

Typical ultrasonic frequencies range between 200kHz and 25MHz and do not, in general, represent significant health hazards.

Several types of sound waves can propagate in solids.

Velocity depends on density and elastic properties of parent material. $V = f \lambda$

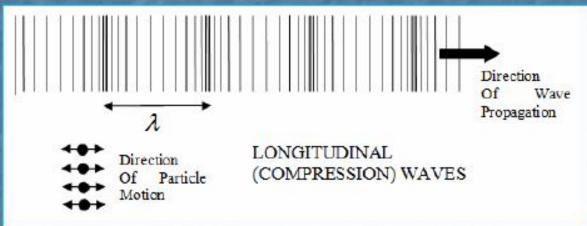
They may be reflected, focussed and refracted.

Longitudinal (compression) and shear (transverse) are two main modes of ultrasonic propagation.

Longitudinal is most commonly used waveform for NDT and occur when ultrasonic beam enters nearly perpendicular to test object surface.

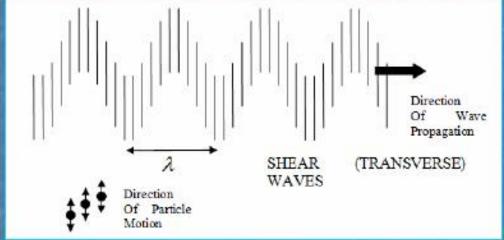
Easily generated and detected.

 Longitudinal waves propagate so that particle vibrations are parallel to direction of wave travel.



Compression wave speed in large body is given by: $V_c = \left(\frac{E(1-v)}{\rho(1+v)(1-2v)}\right)^{\frac{1}{2}}$ where *E* is elastic modulus, *v* is Poisson's ratio and ρ is density of material.

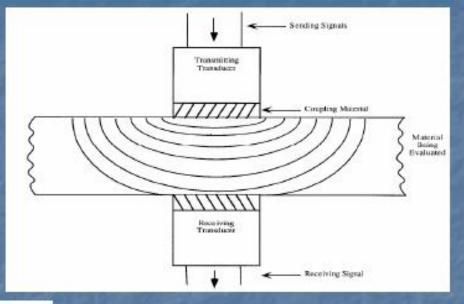
Shear (transverse) waves propagate so that particle motion is perpendicular to wave motion.

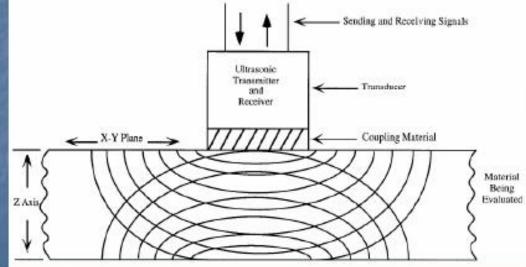


Shear waves are formed when sound enters object surface at moderate angle. Speed given by: $V_s = \left(\frac{E}{2\rho(1+\nu)}\right)^{\frac{1}{2}}$

Ultrasonic waves usually generated by piezoelectric transducers coupled to test object. Sound waves generated when electrical signal is transformed to mechanical vibrations. Receiver converts the vibration back into electrical signals. Two modes of transmitting and receiving signals are used.

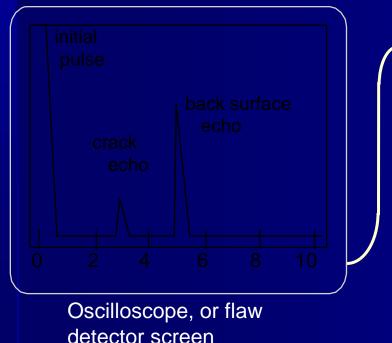
Pitch-catch (through transmission) testing.

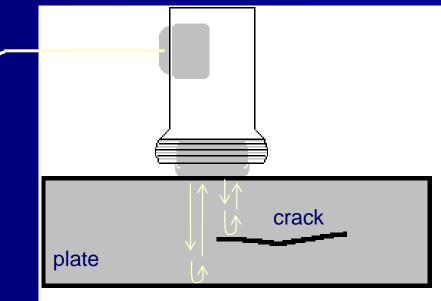




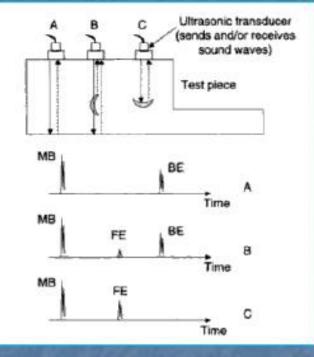
Pulse Echo testing.

Ultrasonic Inspection (Pulse-Echo)

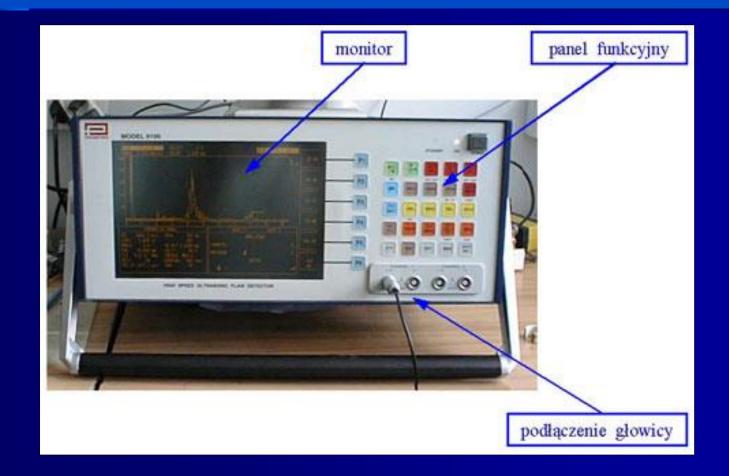




- Ultrasonic signals may be displayed in several formats (or scans) to aid interpretation.
- A-scan is most common form of display.
- Unfaulted object Main Bang (MB) and Back Echo (BE).
- Faulted object flaw location indicated by time of Flaw Echo (FE).



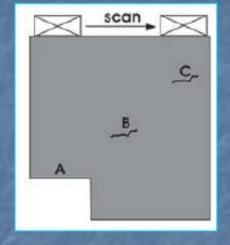
Sensitive to flaw orientation.

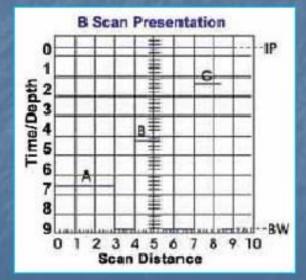


B-scan presents a profile (crosssectional) view of test specimen.

 Time-of-flight displayed on vertical axis and linear transducer position on horizontal axis.

Line A shows reduced thickness and lines B and C indicate flaws B and C indicating depth and length. IP is initial pulse and BW is backwall reflection.

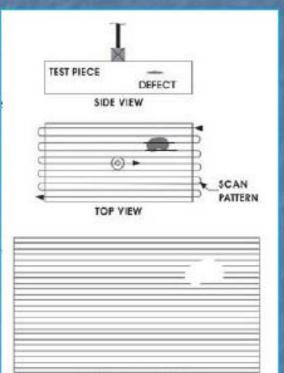




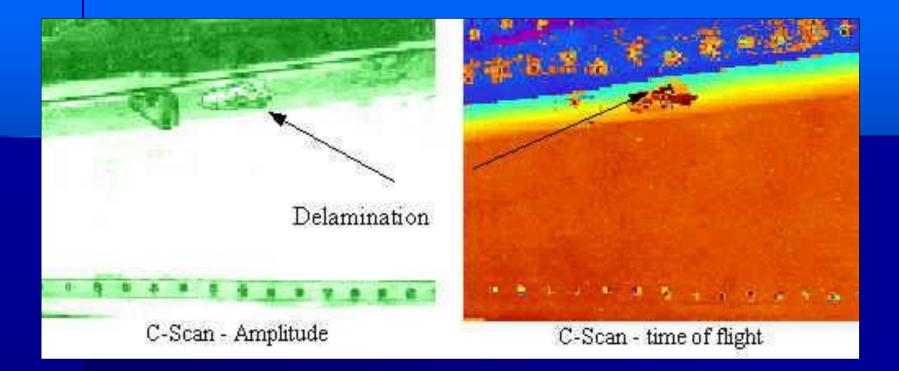
C-scan provides a plan-type view of location and size of test specimen features.

Produced with automated data acquisition system, such as computer controlled immersion scanning system.

Relative time-of-flight displayed as shade of grey or colour indicating location and size of flaw.



C-SCAN PRESENTATION



2-D + colour

Both surface and subsurface anomalies may be detected in relatively large objects. Wide range of metallic and non-metallic materials and thicknesses may be inspected. Information may be obtained regarding flaw size, orientation and location. Portable. Variety of presentation formats. Sensitive to small defects. Little part preparation.

At least one surface must be accessible. Rough surfaces cause problems – couplant required. Highly sensitive to beam and defect orientation. High degree of skill required to interpret. Can be impractical to inspect complex shapes.

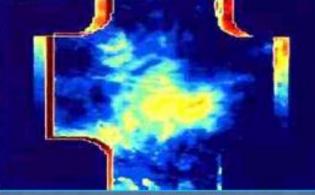
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Thermography and Shearography

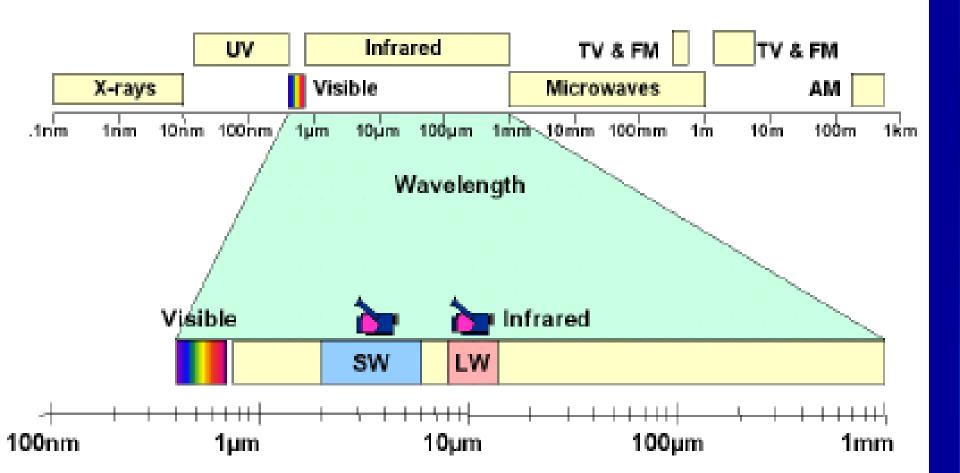
Thermography works on the principle that some anomalies alter the heat flow in the material causing localised temperature differences. Shearography detect stress concentrations around flaws using image-shearing camera which takes images of part in undeformed and deformed states. Test piece illuminated by expanding laser beam.

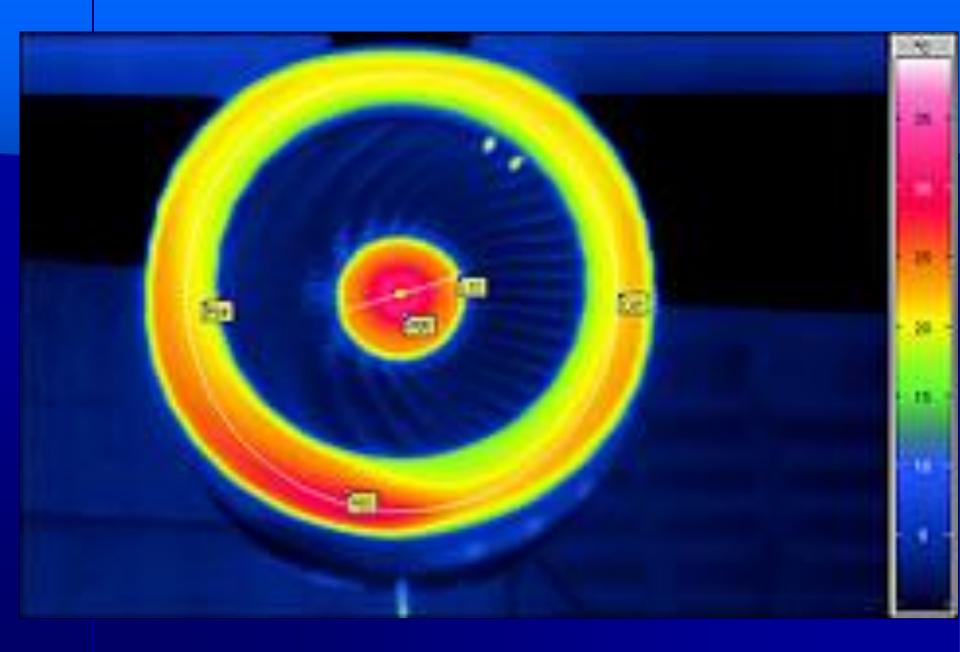


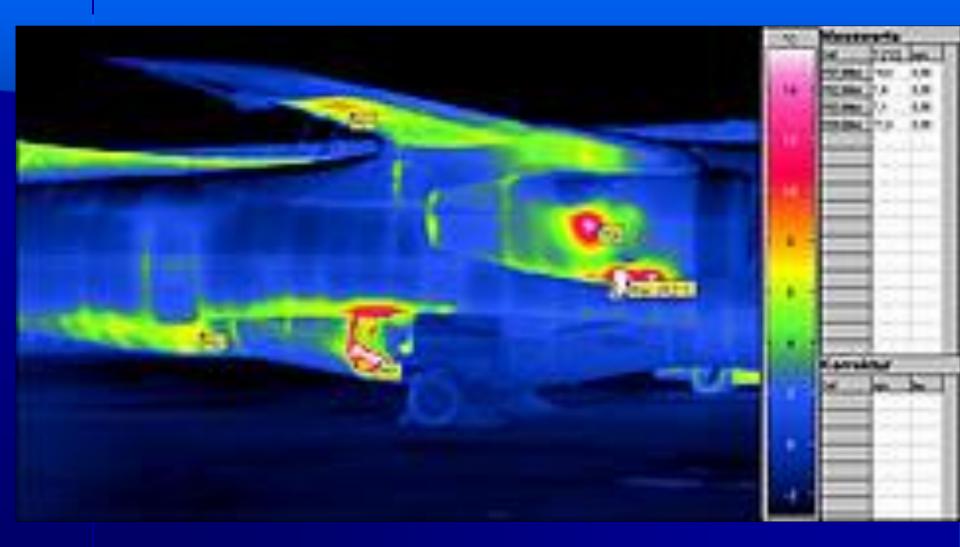


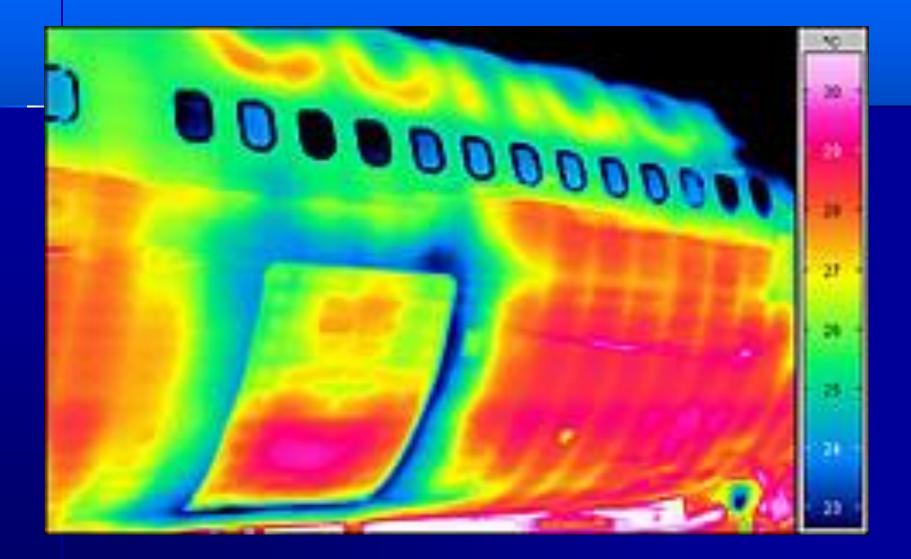
THERMOGRAPHY

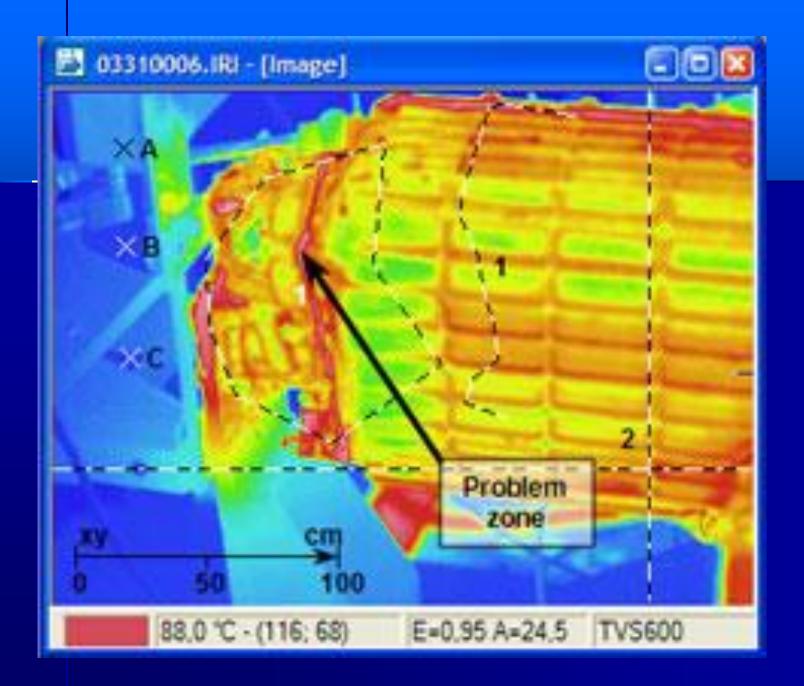
Infrared thermography (IRT), thermal imaging, and thermal video are examples of infrared imaging science. Thermographic cameras detect thermograms. Since infrared radiation is emitted by all objects above absolute zero according to the black body radiation law thermography makes it possible to see one's environment with or without visible illumination.



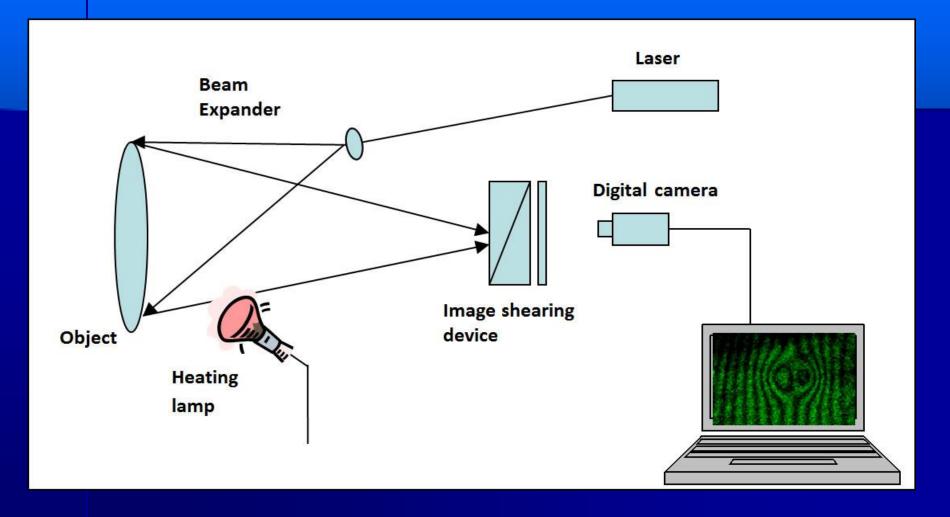




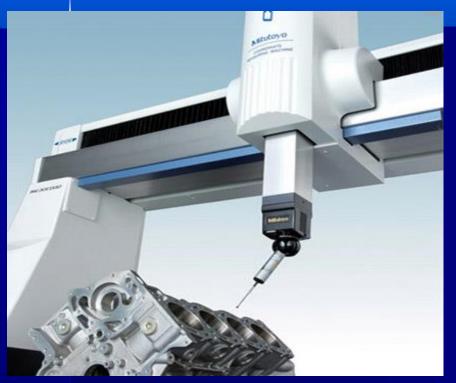




SHEAROGRAPHY



MEASUREMENTS OF DIMENSIONS



http://www.mitutoyo.co.uk/media/wysiwyg/landingpages/cmm.jpg

DIRECT METHODS



http://img.directindustry.com/images_di/photo-g/optical-sensor-for-coordinatemeasuring-machine-cmm-5693-2649407.jpg

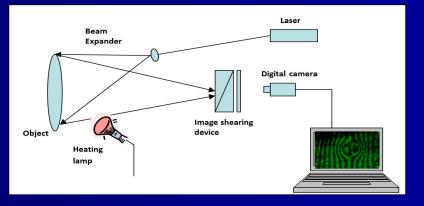
•OPTIC MEASURES

SUMARY: SURFACE METHODS

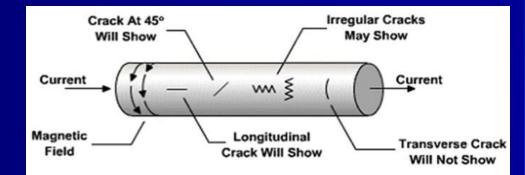
OPTIC/VISUAL



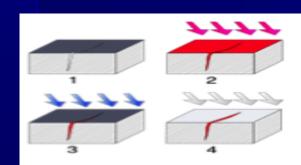
SHEAROGRAPHY



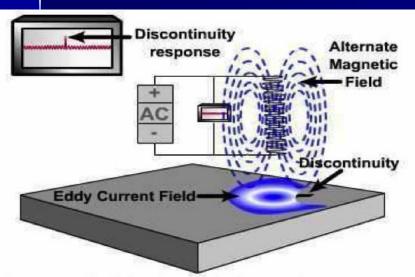
MAGNETIC



LIQUID PENETRANT



SUMMARY: UNDERSURFACE METHODS



Senses conductivity changes in the test piece.

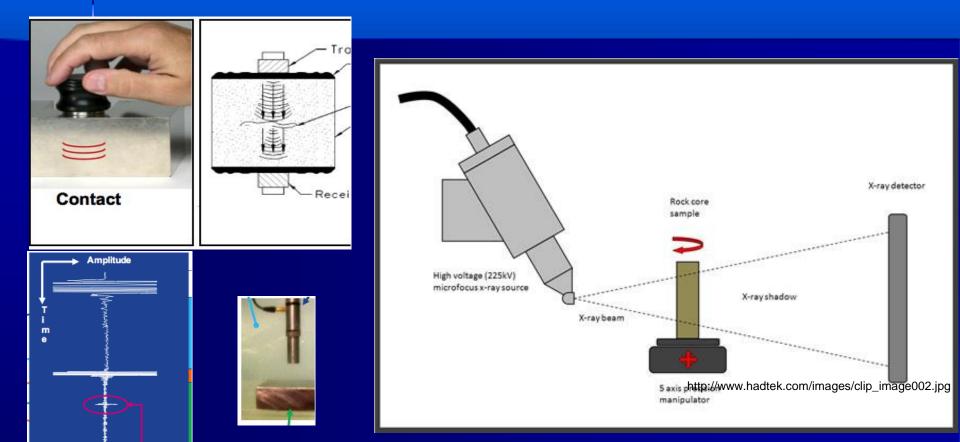
EDDY CURRENT





rysynki pochodzą z udostępnionych slajdów z przedmiotu "Materiały lotnicze"

VOLUME INTERIOR METHODS

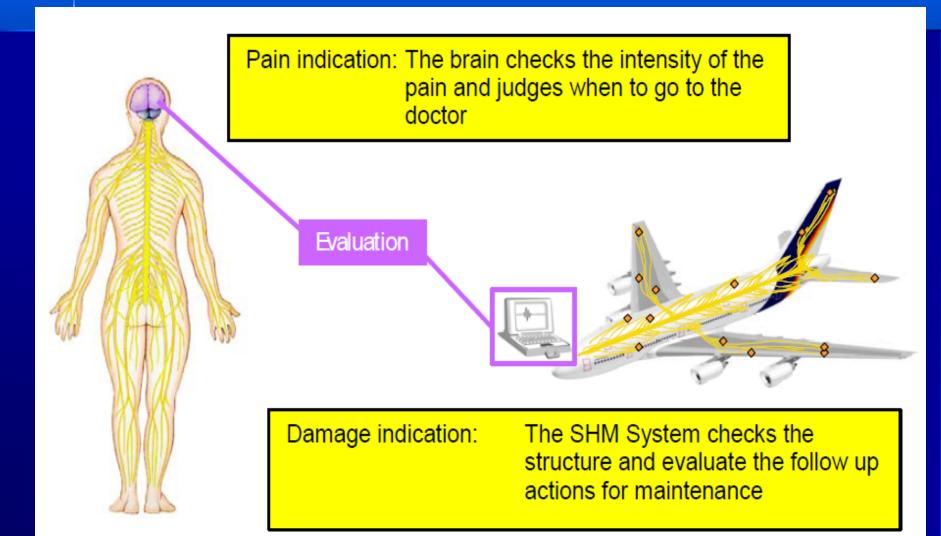




RADIOGRAPHY

SHM – Structural Health Monitoring

SHM



In which way could Nanotechnology help Airbus to support SHM Application Scenarios?

- sensor material
- sensor design
- multi-functional materials
- multi-functional coatings
- signal transfer
- ٠...

Marit Billing

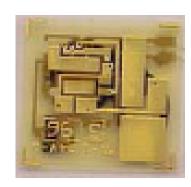
SENSORS



Piezoelectric



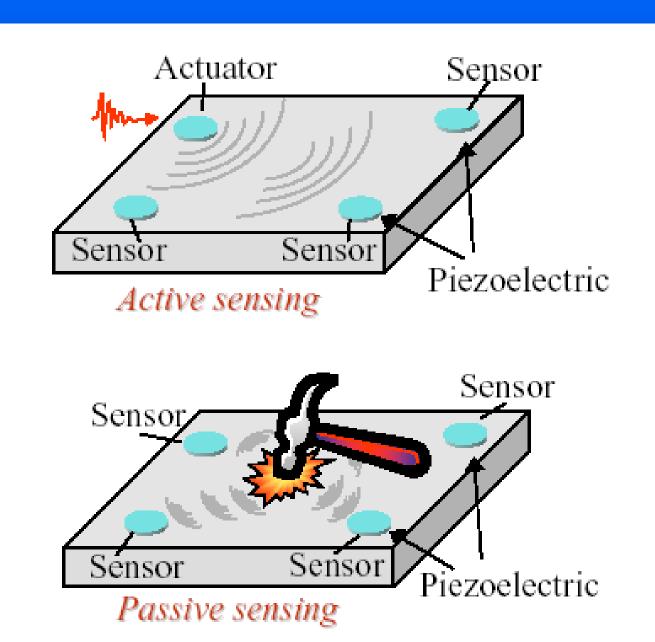
Fiber-optic







Strain-gages



http://www.acellent.com; Acellent Technologies, Inc.,

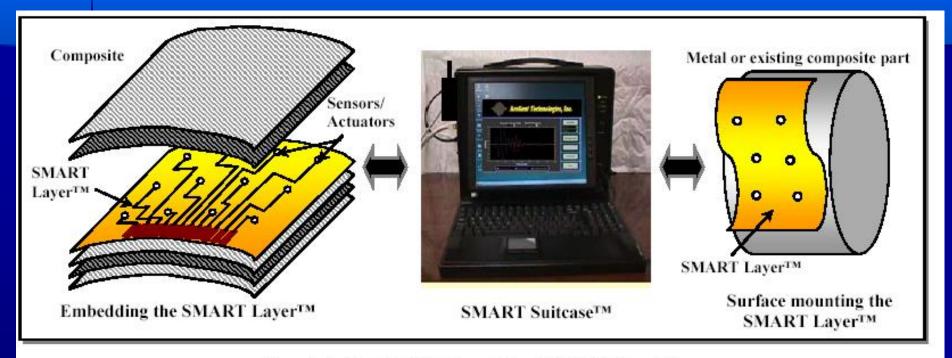


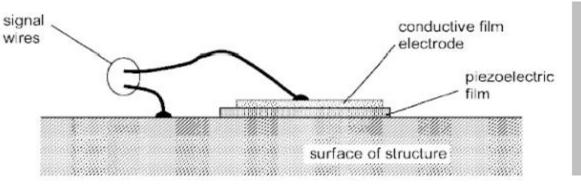
Figure 1: Acellent's SMART Layers™ and SMART Suitcase™.

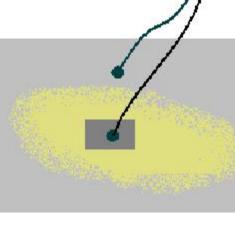
SMART[™] stands for Stanford Multi-Actuator-Receiver Transduction.

Piezo-electric Paint

- Nano Powder of lead-zirconate-titanate (PZT)
- Conversion of mechanical to electrical energy
- Aircraft Surface as a large ultrasound transducer for detection of
 - Impact
 - Vibration
 - Defects









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Piezo Foils

- Nano Powder of lead-zirconate-titanate (PZT)
- Flexible and thin sensors / actuators
- Small contact wires
- Integration of many sensors
- Multi-sensor systems
 - Multi- layer / element sensors / actuators
- High sensitivity
- Broad frequency range (0,001 Hz to several GHz)



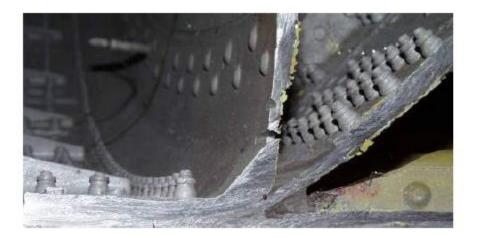


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Metal Structure Application Scenarios

Detection of...

- Cracks (localisation, size)
- Crack growth (localisation, size)
- Accidental damage (localisation, intensity)
- Corrosion (localisation, severeness)
- Loads/Strain (localisation, intensity)









REDUKCJA MASY DLA ELEMENTÓW KOMPOZYTOWYCH

Ważne jest pozyskanie informacji w przypadku struktur kompozytowych o fizycznych własnościach, uszkodzeniach, uderzeniach i rozkładach naprężeń i odkształceń podczas użytkowania SP. Informacje służą do poprawy elementów konstrukcyjnych bądź systemu eksploatacji SP

Dla kompozytów powinno być objętych działaniem SHM 8

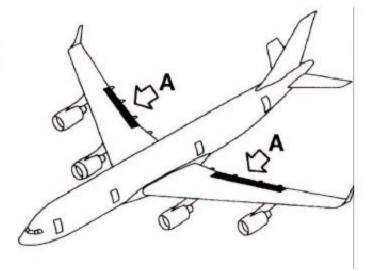
- 1. Impact damage
- 2. Stringer /skin interface failure
- 3. Debonding of CFRP co-bonded parts
- 4. Core / skin sheet debonding in sandwich structures
- 5. Delamination of CFRP-skin layers
- 6. Damage of honeycomb structure
- Detection of missing rivet heads in CFRP structures
- 8. Detection of loads, stress/ strain distribution in CFRP structures.

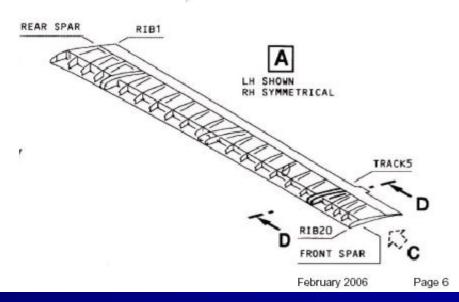
Composite Structure Application Scenarios

Detection of...

- Impact (localisation, intensity)
- Delaminations (localisation, size)
- Debondings (localisation, size)
- Water ingress (localisation, intensity)
- Loads/Strain (localisation, intensity)









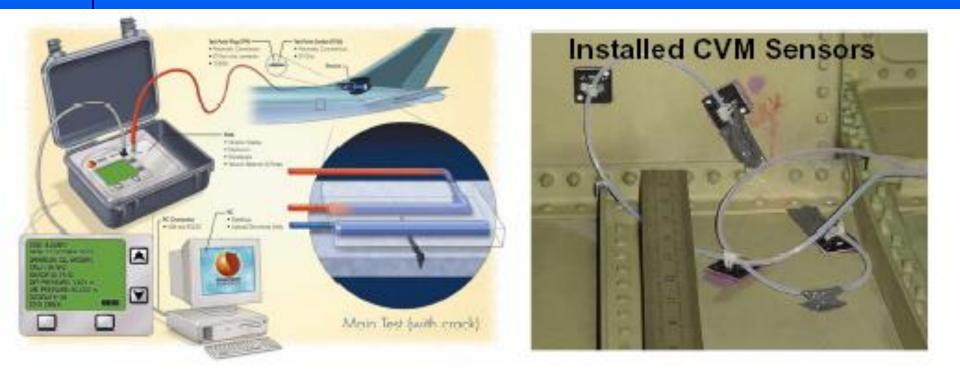
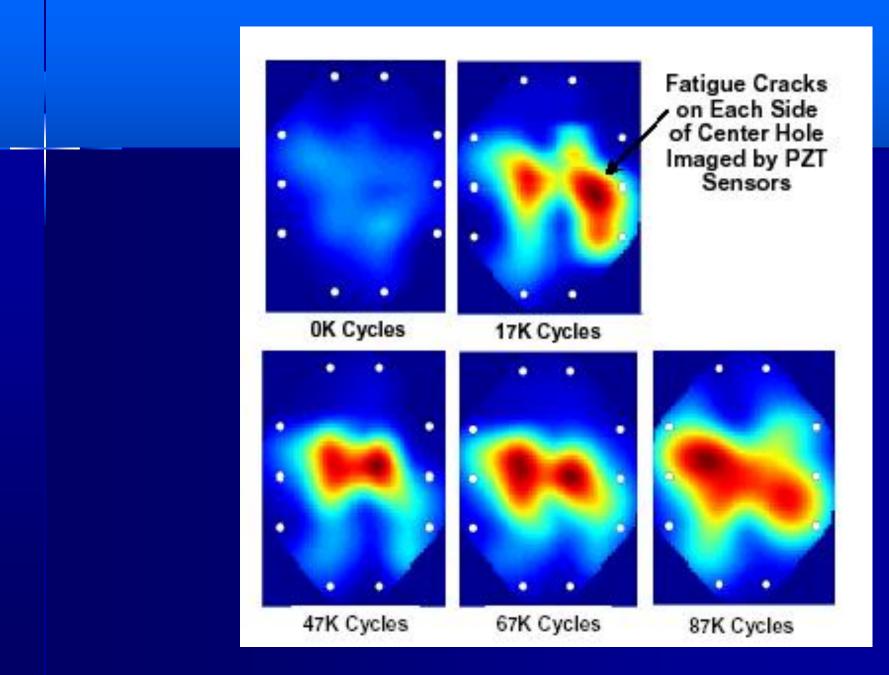


Figure 4: Crack Detection Via CVM System and Aircraft Test Installations of Sensors

Piezoelectric Transducers (PZT)



MONITORING OF A COMPOSITE REPAIR PATCH

